



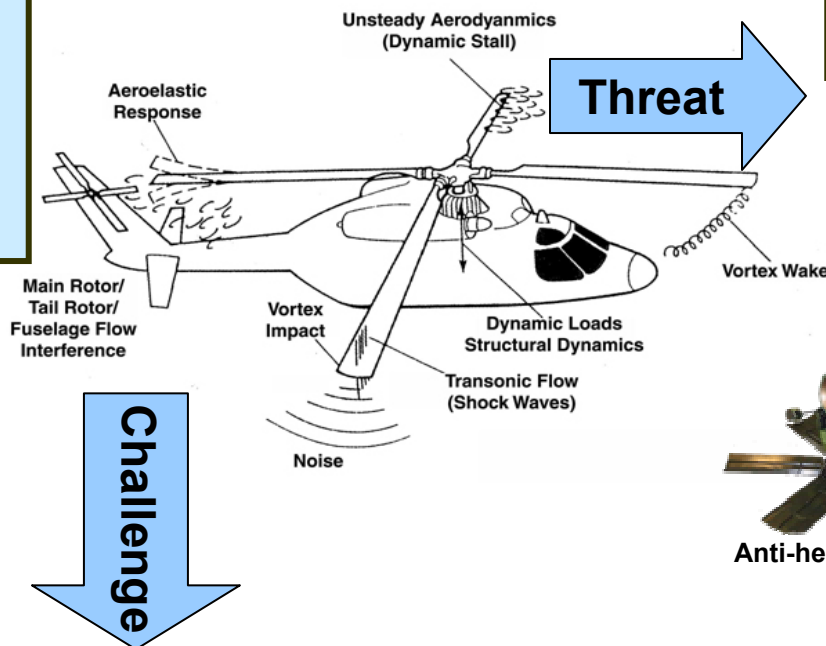
# HELICOPTER QUIETING PROGRAM

**BAA Released: 11 March 2004**

**Proposals Due: 30 April 2004**

**Approved For Public Release  
Distribution Unlimited**

Several complicated mechanisms contribute to the helicopter noise signature



This can be exploited by the enemy



Cueing of air defense systems



Anti-helicopter mine

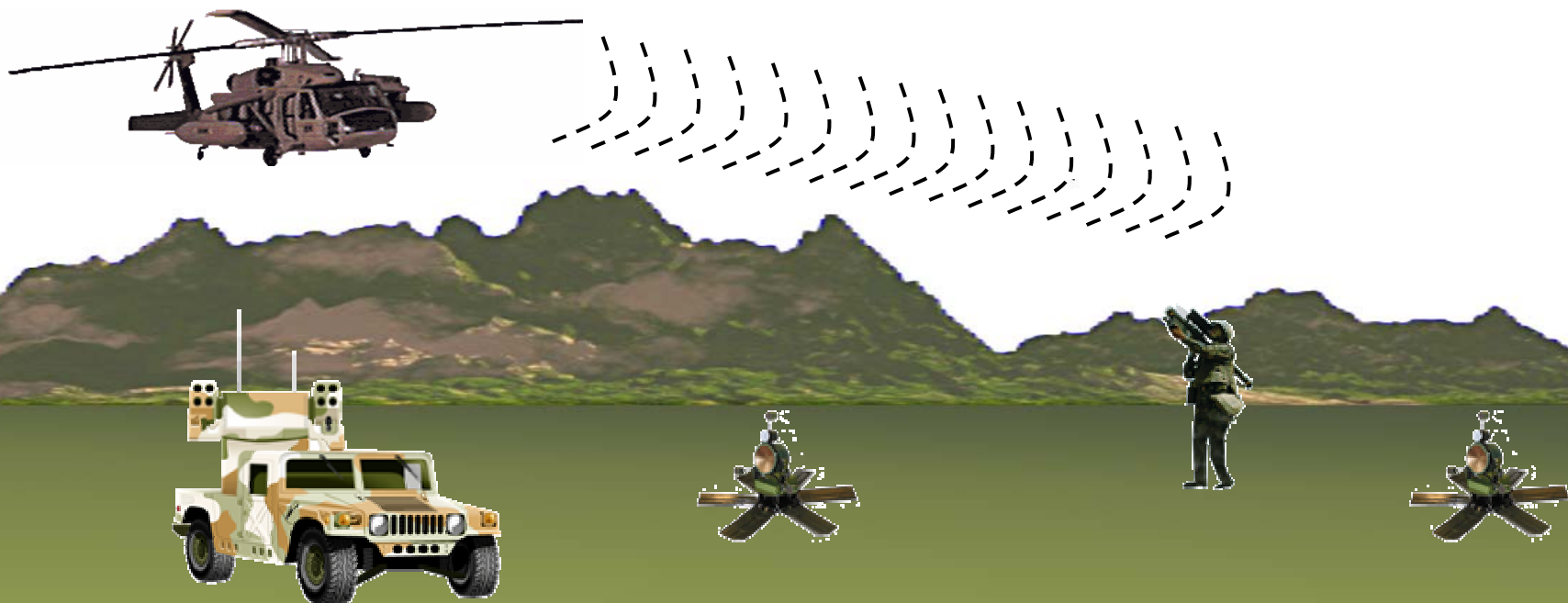
- Wind tunnel testing is very expensive and time consuming
- Most designs are based on empirical data - rotor designs are evolutionary rather than revolutionary
- Current SOA in modeling does not capture all of the physical effects in complex rotational flow
- Most “obvious” methods to reduce noise have a performance penalty



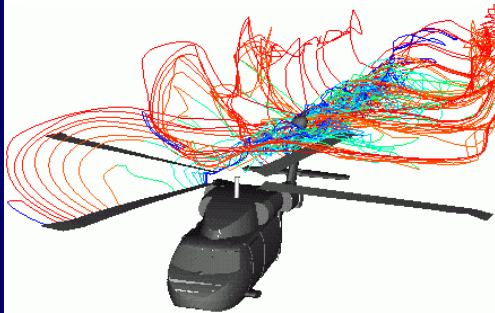
# Program Objective



**Reduce acoustic signature of a helicopter to minimize the probability of detection.**



**End of Program Goal: Order-of-magnitude reduction of low-frequency in-plane acoustic signature without a reduction in performance.**



## Create a new capability

- Develop physics-based, predictive design tools by leveraging high-end CFD techniques
- Validate models using existing data from both wind tunnel and field tests
- Advisors from government and industry provides guidance to ensure applicability of new tool

Phase 1



## Develop a new design

- Design new blades that significantly reduce the low-frequency in-plane acoustic signature
- Verify design through wind tunnel testing

Phase 2

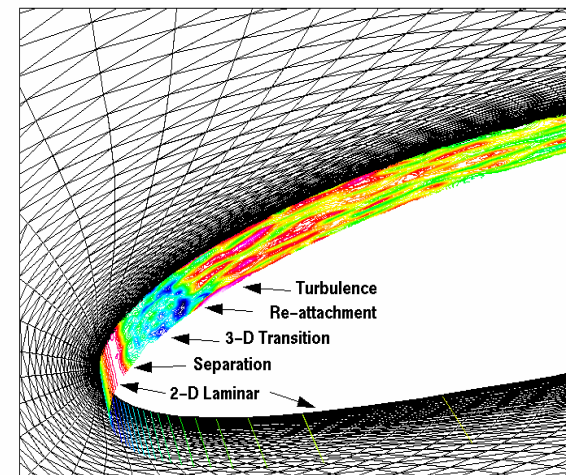


## Flight testing of new design

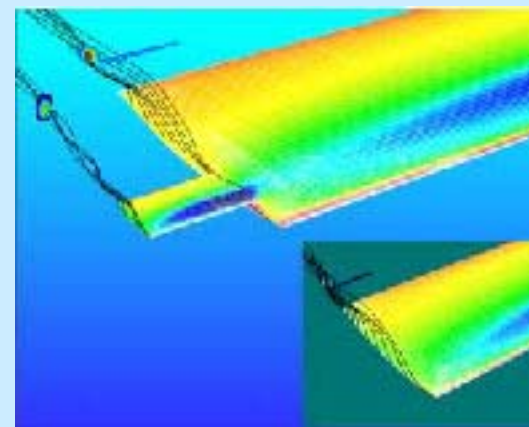
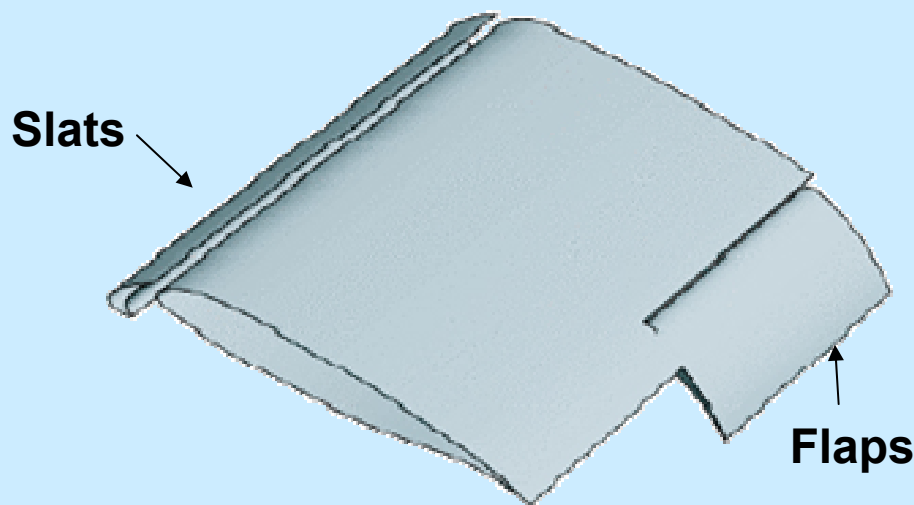
Phase 3

# Possible Approach

- Exploit recent advances in computational fluid dynamics to develop a **physics-based** design tool that can **predict** the detailed blade loading and performance **regardless of blade shape**
- Utilize the tool to develop **novel blade designs** with vastly improved acoustic characteristics



## POSSIBLE NOVEL BLADE DESIGNS



Unique tip designs





# Tentative Schedule & Milestones

